INTERACTION BETWEEN LARGE BLAST AND TARGETS CAN RARELY ONLY BE DIRECTLY STUDIED, DUE TO COST AND PRACTICABILITY CONSIDERATIONS. BLAST TESTS USING REDUCED-SCALE HIGH EXPLOSIVE CHARGES REPRESENT AN ATTRACTIVE ALTERNATIVE. THE FIRST NECESSARY STEP CONSISTS IN STUDYING BLAST PROPAGATION IN FREE-FIELD AT THE CONSIDERED SCALES. THE SECOND STEP FOCUSES ON THE DETERMINATION OF THE BLAST LOAD AROUND VARIOUS TYPES OF REFERENCE OBSTACLES, IN ORDER TO PROVIDE A CRITICAL INPUT FOR NUMERICAL SIMULATION. THIS APPROACH ALSO AIDS TO BUILD SIMPLIFIED MODELS ALLOWING FASTER RISK ASSESSMENT FOR GOVERNMENT AGENCIES.


TWO RESPECTIVE SERIES OF CHARGES WERE DETONATED AT DISTANCES BETWEEN 0.6 AND 3.5 m/kg 1/3 FROM THE HEMI-CYLINDER, IN ORDER TO ASSESS ITS INFLUENCE ON THE WAVE REFLECTION STRUCTURE AND THE RESULTING BLAST LOAD. HIGH-SPEED IMAGES ENABLED THE TRIPLE AND CONTACT POINTS TRACKING ON THE OBSTACLE. FINALLY, THIS PROJECT ILLUSTRATES AN INNOVATIVE METHODOLOGY NOT ONLY TO ASSESS THE BLAST LOAD ON A CONVEX STRUCTURE, BUT ALSO THE POTENTIAL DOWNSTREAM PROTECTIVE EFFECTS OF SUCH A STRUCTURE USED AS A BARRIER.

KEYWORDS: BLAST WAVE, HEMI-CYLINDRICAL STRUCTURE, DETONATION, PLASTIC EXPLOSIVES, REDUCED-SCALE CHARGES, EFFECT ON INFRASTRUCTURES, EXPERIMENTS, CONVEX, MACH REFLECTION.

1 INTRODUCTION

THE INSTITUTE FOR RADIOLICAL PROTECTION AND NUCLEAR SAFETY (IRSN) IS A FRENCH PUBLIC INSTITUTE WITH INDUSTRIAL AND COMMERCIAL ACTIVITIES, PLACED UNDER THE JOINT AUTHORITIES OF THE MINISTRIES OF DEFENCE, ENVIRONMENT, INDUSTRY, RESEARCH, AND HEALTH. IRSN IS ENTRUSTED, AMONG OTHERS, TO ASSESS AND CONDUCT RESEARCHES IN THE AREA OF THE PROTECTION OF NUCLEAR FACILITIES AND TRANSPORT OF RADIOACTIVE AND FISSILE MATERIALS AGAINST ACCIDENTAL AND MALICIOUS ACTS. IN THIS CONTEXT, IRSN ESTABLISHES PROJECTS AND STUDIES TO IMPROVE ITS KNOWLEDGE OF BLAST CHARACTERISTICS AND WEAPONS EFFECTS.

In 2006, IRSN designed and built an experimental set-up to achieve non-destructive shock wave propagation studies on a small scale [1], [2]. This set-up is composed of a modular table, sensors and targets able to perform the detonation of solid explosives up to 64 g of TNT equivalent, representing an alternative to the gas mixture detonation propagation configuration for small-scale tests [3], [4]. Blast generated by 50 g TNT equivalent hemispherical Hexomax® charges was consequently characterized. To evaluate the effect of scale, a joint study between IRSN and ISL was initiated in 2017, based on a dedicated ISL blast pad replicating the IRSN table at a double scale: 400 g TNT equivalent hemispherical Hexomax® charges are being investigated.

Based on literature and IRSN previous experimental results [5], [6], this document proposes innovative abacuses (Model S) to predict the three-dimensional peak overpressure on the expansion side of a convex non-deformable structure (simplified shape based on an actual transport cask or industrial facility and different from the more common box-like obstacle) depending on the reduced distance to the obstacle. Data collected during the 2018 joint IRSN-ISL campaign [6] on 400 g TNT equivalent charges are consequently compared to the model predictions. In addition, the fine structure of the shock wave bypassing the convex surface is studied using a high speed imaging setup based on direct lighting. The Mach stem height is measured in terms of triple point radius and angle of observation on cylinder. This height versus angle curve serves to establish the triple point time trajectory over the convex shape.

2 EXPERIMENTAL SETUP

2.1 IRSN setup

The IRSN blast table has been principally designed to study shock waves reflection phenomena and interaction with different non-deformable structures [2]. It measures 1.6 × 2.4 m and features an array of mounting holes that facilitates the placement of modular 0.4 × 0.4 × 0.05 m wooden plates and pressure transducers (Fig. 1). Explosive charges are installed and ignited on a dedicated reinforced steel table plate to generate a hemispherical blast wave using a Davey-Bickford SA 4201A detonator. Experimental campaigns were performed at the ArianeGroup’s research center located near Paris (Vert-le-Petit, France).

![IRSN blast table](image)
A wooden hemi-cylindrical structure was fixed on the table surface. The 0.4 m diameter hemi-cylinder has the same length (1.6 m) as the table in order to avoid lateral bypass of the shock wave (see Fig. 2). The target is sufficiently rigid to withstand repeated blast loading without damage nor deformation. Five pressure gauges separated by 30° can be installed on the target surface along three vertical lines facing the explosive charge. For this campaign, five pressure transducers were mounted on elastic supports and their signals were recorded at 500 kHz by a Nicolet Genesis data acquisition system.

Results presented in this document were acquired for three different d distances (Fig. 3): 0.4, 0.8 and 1.6 m (Table 1) [5]–[7].

Figure 2: View of the wooden hemi-cylinder lying on the table with holes for sensor positioning.

Figure 3: Diagram of the experimental configuration involving a hemi-cylinder.

Table 1: Hemi-cylinder metrology specifications.

<table>
<thead>
<tr>
<th>Angle μ</th>
<th>ISL PCB</th>
<th>ISL Kulite</th>
<th>IRSN Kistler</th>
</tr>
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<tr>
<td>30°</td>
<td>x</td>
<td>x</td>
<td>x</td>
</tr>
<tr>
<td>60°</td>
<td>x</td>
<td>x</td>
<td>x</td>
</tr>
<tr>
<td>90°</td>
<td>x</td>
<td>x</td>
<td>x</td>
</tr>
<tr>
<td>120°</td>
<td>x</td>
<td>x</td>
<td>x</td>
</tr>
<tr>
<td>150°</td>
<td>x</td>
<td>x</td>
<td>x</td>
</tr>
<tr>
<td>Lateral offset (mm)</td>
<td>0</td>
<td>50</td>
<td>0</td>
</tr>
<tr>
<td>d (m)</td>
<td>3.2</td>
<td>0.4/0.6/1.6</td>
<td>0.6/1.6</td>
</tr>
</tbody>
</table>
2.2 ISL setup

ISL developed a dedicated outdoor blast pad (Fig. 4): explosive charges are detonated in a factor two up-scaled version of IRSN test configuration. Each explosive charge was installed on a dedicated reinforced steel ground plate to generate a hemispherical blast wave. Based on IRSN previous work, ISL also designed a factor two upscaled hemi-cylindrical obstacle (Fig. 5). The results obtained for two target configurations are described in this document:

- “long” hemi-cylinder composed of 4 segments (800 mm diameter and 3.2 m total length),
- “short” single instrumented (5 lines of instrumentation ports) hemi-cylinder (800 mm diameter and 0.8 m length).

![Figure 4: ISL blast pad.](image)

![Figure 5: ISL setup with the “long” (a) and “short” (b) hemi-cylindrical obstacles.](image)

Presented results were recorded at a distance \( d \) of 3.2 m at ISL scale (Table 1). Different types of pressure transducers were mounted on polypropylene supports inserted in the hemicylinder ports each separated by 30°. All data were recorded using a Transcom system running at 2 MHz.

2.3 Pressure sensors positions

The authors would like to draw the reader’s attention to the fact that all distances presented in the rest of this document correspond to IRSN scale. ISL distances and times have all been downscaled for direct comparison.
Table 1 presents the position of all sensors for hemi-cylinder tests at both scales (reduced distances $Z$ ranging between 0.57 and 4.6 m.kg$^{-1/3}$ for a TNT equivalent spherical charge). IRSN hemi-cylinder centre line was equipped with five Kistler 603B sensors every 30°. For the ISL obstacle, five PCB 113B28 sensors populated the centre line whereas a 50 mm distant (ISL scale) line included five Kulite XT190 sensors (Table 1).

2.4 Explosive charges

The explosive charges consisted of Hexomax® hemispheres, initiated from the bottom using an electrical detonator (Davey-Bickford SA-4201 at IRSN or RP83 at ISL). For these tests, the used masses were 41.6 g (IRSN) and 333 g (ISL) of Hexomax® (50 and 400 g respectively in TNT overpressure equivalent). Each charge was placed directly either on the table (IRSN) or the blast surface (ISL) as seen in Fig. 6. Charges characteristics are reported in Table 2.

![Hexomax® charges. (a) IRSN; and (b) ISL.](image)

Figure 6: Hexomax® charges. (a) IRSN; and (b) ISL.

<table>
<thead>
<tr>
<th>Configuration</th>
<th>IRSN</th>
<th>ISL</th>
</tr>
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<tbody>
<tr>
<td>Mass (g)</td>
<td>41.6</td>
<td>333</td>
</tr>
<tr>
<td>Diameter (mm)</td>
<td>46.6</td>
<td>94</td>
</tr>
<tr>
<td>Density (g/cm$^3$)</td>
<td>1.58</td>
<td>1.54</td>
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<tr>
<td>Igniter</td>
<td>SA-4201</td>
<td>RP83</td>
</tr>
<tr>
<td>Pressure TNT eq. (g)</td>
<td>50</td>
<td>400</td>
</tr>
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</table>

3 EXPERIMENTAL RESULTS AND ANALYSIS

3.1 Overpressure evolution at the hemi-cylinder surface

Overpressure evolutions along the IRSN hemi-cylinder central line versus angle of observation are presented in Fig. 7 for the three different $d$ distances: 0.4, 0.6 and 1.6 m. Each represented point corresponds to the average of three experimental measurements. Peak overpressure values decrease with the angular position on the obstacle as the compression decreases before the transition to the expansion on the backside. A similar trend seems to appear for all distances $d$ after 90°. Mach transitions angles are estimated at 28°, 30° and 44° respectively for 0.4, 0.6 and 1.6 m according to Kinney and Graham [8].
Fig. 7: Overpressure evolution on the central line for the 0.4, 0.6 and 1.6 m distances at IRSN scale [5], [6], [8], [9].

Free-field overpressures for an incident wave according to the Kinney and Graham theory [8] are also represented on the same graph in dashed line. Another simplified model [8]–[10] represented in black lines is based on TM5-1300 [9]: reflection coefficients are determined for planar targets and corresponding incident angle and overpressure. These data are only plotted for the reflection face of the target, as the model is not able to predict the wave diffraction on the opposite side.

On the blast-exposed face, average measurements are in reasonable agreement with the model (up to 50% discrepancy). The limited number of angular positions for the sensor however prevented us from determining the position of the classic “knee” on the pressure evolution.

At 60°, after the formation of the Mach stem, the peak reflected overpressures are greater than those predicted by the simplified model: this can be explained by the Mach stem curvature above the convex shape (Fig. 12).

At 90°, at the beginning of the expansion zone, the measured overpressures are slightly higher than the free-field values.

At 150°, overpressures behind the obstacle are at least 55% lower than for free-field at the same position, demonstrating the blast mitigating effect of the hemi-cylinder.

3.2 Transmission coefficient

Transmission coefficient versus direct distance to the measurement point is presented in Fig. 8 for IRSN measurements. This coefficient is defined by the ratio between the measured and the predicted free-field [8] overpressures at the considered position (direct distance from the charge centre to the considered position at the hemi-cylinder surface). This consequently corresponds to a classic reflection coefficient on the obstacle front surface. On the expansion side of the obstacle, it illustrates the obstacle’s ability to reduce the overpressure by
diffracting the incoming blast wave. The angle of vision limit is materialized for each configuration by a dashed vertical line in Fig. 8. This definition of the transmission coefficient differs from the classic shock interaction with a hemi-cylindrical shape approach considering a constant reference pressure: previous studies usually focus on 2D cylinders submitted to a steady pressure wave generated by a far-field large-scale blast [11] or inside a shock tube [12], [13]. For this study, the 3D blast wave intensity decreases as it propagates and interacts with the finite hemi-cylinder.

Figure 8: Transmission coefficient evolution along the central line versus direct distance [5], [6].

Transmission coefficient values are comprised between 1.25 and 4.5 before the angle of observation limit, in agreement with literature [9]. After the limit angle, it decreases from 1 down to 0.2.

This approach is based on the direct distance evolution. The deployed distance evolution could be considered as well.

Based on these results, an innovative approach of the transmission coefficient evolution is proposed by analyzing its variation depending on the direct distance with an origin shifted to the beginning of the expansion zone (Fig. 9) and divided by the hemi-cylinder radius to obtain a reduced shifted direct distance (RSDD): in this region of the hemi-cylinder surface (RSDD comprised between 0 and 1.2), corresponding to different angular ranges for each configuration, all values seem to follow a similar trend for the three different values of d/R (2, 3 and 8).

From the observations in Fig. 9, an experimental correlation, eqn (1), was proposed (named Model S) and illustrated in Fig. 10:

\[ Ct = 0.396 \times \text{RSDD}^2 - 1.4385 \times \text{RSDD} + 1.347, \]

(1)

where

- \( Ct \): transmission coefficient (dimensionless);
- \( \text{RSDD} \): reduced shifted direct distance (dimensionless).
Figure 9: Transmission coefficient evolution along the central line versus reduced shifted direct distance (RSDD).

Figure 10: Transmission coefficient evolution – experiments versus Model S.

Finally, Fig. 11 presents IRSN overpressure data collected versus angle of observation in comparison with literature (KG free-field [8] and TM5-1300 [9]) and Model S correlations determined with IRSN results (for the 3 d distances 0.4, 0.8 and 1.6 m). Model S is plotted by solid coloured lines on the angular range of validity (from the critical angle of observation to 180°) and extended on the shock reflection front face in dashed coloured lines.

All ISL scale peak overpressures recorded for short and long hemi-cylinders (d = 3.2 m) using PCB and Kulite sensors are added as green symbols in Fig. 11. The Model S correlation provides a convincing prediction of the 400 g TNT equivalent experimental overpressure.
Figure 11: Overpressure evolution versus angle of observation on the hemi-cylinder central line – experimental data versus Model S.

evolution on the back side of the upscaled obstacle. Results dispersion at 150° can be explained by the expansion phenomenon itself, the proximity of a welding joint next to the sensor positions as well as the difference in the responses of the pressure gauges [14]. This first inter-scale comparison strengthens the pertinence of the Model S conjecture to predict peak overpressure evolution on the expansion side of a cylindrical target exposed to a blast wave for any d/R ratio and charge mass by defining a reduced shifted direct distance. This model completes the TM5-1300 approach for the complex unsteady gas-dynamic interaction phenomena [11].

4 HIGH SPEED IMAGING

Fig. 12 illustrates the evolution of the blast wave interaction with the ISL hemi-cylinder on high-speed images 5.94 ms after a 333g Hexomax® charge detonation at 3.2 m (ISL scale) from the obstacle. Background Oriented Schlieren (BOS) [15] was chosen to enhance the visibility of the shock propagation [6]. BOS image is presented below the original record. At this stage of the propagation, the three characteristic waves (incident, reflected and Mach) are visible, merging on the triple point.

Fig. 13 shows the triple point radius (left) and velocity (right) versus time evolutions [6]. These data were extracted from the high-speed images recorded for the same Hexomax® test. The triple point appears 5 ms after the explosive charge detonation and propagates at an average of 450 m/s between 5 and 7 ms. This technique based on BOS consequently permits an accurate tracking of the triple point and the global structure of the reflected wave on the convex shape.
5 CONCLUSION AND PERSPECTIVES

Blast wave interaction with a convex structure was investigated by IRSN and ISL through a twin reduced scale experimental approach. The effects of hemispherical Hexomax® charges were characterized at the surface of a hemi-cylindrical obstacle. Based on the 42 g charge (50 g TNT equivalent) configuration defined in previous IRSN work, ISL designed a factor 2 up-scaled setup by using 333 g Hexomax® hemispheres (400 g TNT equivalent). Blast wave propagation was analysed through the peak overpressure recorded with different types (piezo-electric and piezo-resistive) of pressure sensors placed at the target surface.

Overpressure evolutions along the IRSN hemi-cylinder central line were determined for three distances to the explosive charge. Results are compared to a reflected face simplified model (TM5-1300 [9] + Kinney and Graham [8]) and free-field side on overpressure (Kinney and Graham [8]). A good agreement is observed between experimental data and the simplified model on the directly exposed face of the obstacle. To predict the blast load on the expansion side in a similar way, transmission coefficients are represented as functions of an innovative parameter: the reduced shifted direct distance (RSDD). By taking into account the geometry of the obstacle and the 3D free-field pressure decay, this physical conjecture (Model S) allows the prediction of the peak overpressure evolution on the expansion side of
the ISL hemi-cylinder at double scale. An experimental campaign conducted in January 2020 confirmed the extension of this conjecture to other charge masses and d/R ratios [16]. This model will also be extended to other blast parameters such as positive phase impulse and duration. This type of tool will prove to be useful to perform structural finite element analysis or to take part in the design of real critical infrastructure by government agencies in addition to other classic tools [17] able to predict average loading on blast exposed surfaces.

REFERENCES


