

Identification of a canonical configuration for a quantitative analysis of transport phenomena in electric machines based on entropy generation

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Abstract

The demand for higher power density at low costs leads to an increasing importance of thermal management in electrical machines. Different approaches are employed to develop cooling strategies. However, a truly quantitative analysis is barely used, partly due to the huge data quantity generated by the flow simulation. In many cases only visual postprocessing under consideration of a few variables is implemented. The present paper provides an overview about a method suitable for investigating transport phenomena induced by forced convection. Different configurations are retained to check the generality of the approach. Based on the second law of thermodynamics, the entropy generation in the fluid field is considered. As an example of external flow, a semi-infinite plate is examined. For internal flows, two rotating, concentric cylinders are then analysed. Finally, transport processes in the flow between these cylinders located in a rectangular domain (Ω) are analyzed. In addition, the outer cylinder has two openings along the main stream direction. The straight flow imposed at the inlet velocity of the domain (Ω) is influenced by the rotational motion of the inner cylinder as well as by the opening angle (β) at the outer cylinder and the radius ratio between the two cylinders (II). The thermal boundary conditions are isothermal or isoflux and isothermal or adiabatic for the inner and outer cylinder, respectively. In a first step, numerical calculations of the Nusselt number (Nu)



are discussed. This methodical approach can be a first step to analysing transport phenomena under consideration of physically-based indicators.

Keywords: transport phenomena, entropy generation, rotating cylinder, heat transfer, cooling, electrical machines.

1 Introduction

Cooling by forced convection has a wide range of application in engineering. To increase the lifetime of electric devices, the objective is to maintain the operating temperature at a safe level. Inside an alternator, fan blades generate a pressure gradient, due to the rotational motion. Air from the engine bay streams through the alternator and thereby increases the enthalpy transport. With the help of computational fluid dynamics (CFD) the flow field can be analyzed. The main objective is a better design while considering certain constraints.

In [1] Thévenin and Janiga showed how to combine Evolutionary Algorithms with CFD (an approach also called CFD-O). In chemical engineering the Nusselt number (Nu) can be used as an efficiency objective to design different kinds of system configurations as presented by the VDI Heat Atlas [2]. The fundamental work of Bejan [3] introduced a new way to analyse different processes based on the second law of thermodynamics. The consideration of exergy destruction due to conduction and fluid friction offers a new way to look at heat transfer problems. Moreover, the entropy generation is a postprocessing value provided by a volume integral over the domain of interest. Fowler and Bejan [4] showed optimal sizes of bodies for external flows with the objective of entropy generation minimization (EGM). Herwig and Schmandt [5] used the second law analysis (SLA) to determine the drag coefficient for external flows with the help of the volumetric entropy generation in the flow field. In contrast to external flows, SLA-approaches for internal flows are widely available. Wenterodt and Herwig [6] and Herwig *et al.* [7] showed that an optimal Reynolds number (Re) exists for a heated pipe considering EGM. Ko and Tink [8] and Ko [9] investigated EGM for rectangular duct geometries with longitudinal ribs. Şahin [10] varied duct geometries, gave an analytical formulation and estimated the best geometry based on EGM. Aziz [11] summarized analytical studies of entropy generation in Couette flow for different kinds of boundary conditions. Mahmud and Fraser [14] and Mahian *et al.* [17] presented an analytical solution for the entropy generation between two concentric rotating cylinders. Mirzazadeh *et al.* [13] extended the analytical solution for non-linear viscoelastic fluids. Yilbas [18] computed temperature rise and entropy generation due to conduction and fluid friction.

Based on the same geometry of two rotating, concentric cylinders, the present authors included openings along the outer cylinder. Hence, it is more accurate to speak of an outer sleeve instead of a cylinder for this case. The idea is to involve different types of transport phenomena for testing physically-based indicators, e.g. entropy generation. For this purpose a configuration is needed, which is able to involve a variety of flow fields dominated by different heat transport mechanism.

2 Physical model

Based on conceptual considerations in the fields of aerodynamic and transport phenomena, the authors developed a canonical configuration that will be systematically used for further analysis.

2.1 Conceptual considerations

Figure 1 shows two fundamental cases for aerodynamic studies. Because of the employed boundary conditions, velocity δ and thermal δ_T boundary layers develop along the bodies.

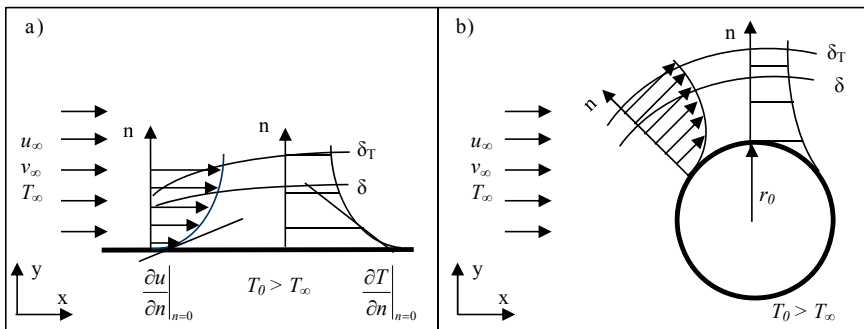


Figure 1: Boundary layer formation for a) semi-infinite flat plate and b) cylinder.

Near the wall, velocity and temperature gradients are high. A high Nusselt number results in an increasing velocity gradient. For the semi-infinite flat plate in fig. 1a), the Reynolds Analogy can be used to relate the flow parameters to the thermal parameters. In this case, Nu can be formulated as:

$$Nu = \frac{hL}{k} = \frac{c_f}{2} Re \quad (1)$$

where h is the local heat transfer coefficient, L is the characteristic length and k is the thermal conductivity of the fluid. Re is the Reynolds number and c_f is the skin friction coefficient, defined as:

$$Re = \frac{\rho u_\infty L}{\mu} \quad (2)$$

$$c_f = \frac{2\tau}{u_\infty^2 \rho} = \frac{2\mu}{u_\infty^2 \rho} \left. \frac{\partial u}{\partial n} \right|_{n=0} \quad (3)$$

Thereby ρ defines the density of the fluid, μ the dynamic viscosity, u_∞ the ambient velocity and τ the shear stress. In such cases, the velocity gradient could

be a criterion to characterize heat and mass transfer. The transport of energy in form of heat is often described by the kinetic approach:

$$Q = hA(T_w - T_\infty). \quad (4)$$

From eqs. (1) and (4) it is obvious that the transport is related to the gradients near the wall and the temperature difference $\Delta T = T_0 - T_\infty$. This results in two different kinds of processes:

- near-wall processes (local) and
- enthalpy transport on a global scale.

Extending now the geometry from fig. 1b) with a vertical plate on the left hand side, the flow profile will change like described in fig. 2.

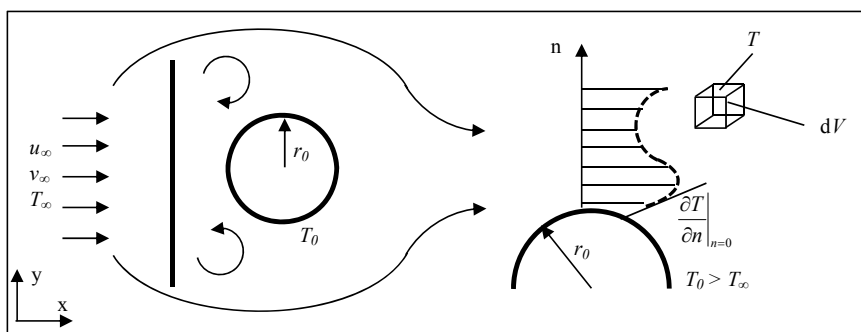


Figure 2: Temperature boundary layer profile for a specific case with interaction.

In comparison to the cases in fig. 1, a reference temperature T in fig. 2 is difficult to determine due to complex processes in the flow field. Based on the first law of thermodynamics, the change of enthalpy due to heat transfer can be calculated with:

$$d\dot{q} = d\dot{h} = c_p \Delta T \quad (5)$$

where h is the specific enthalpy and c_p the specific heat capacity for constant pressure. This small change in the model shows the difficulty of finding a useful reference temperature T . Fig. 3 illustrates another variation starting from a basic case of aerodynamics. In fig. 3a) the boundary layer developing along a flat wall interacts with a rotating cylinder placed inside a wall recess. When increasing the angular velocity ω_0 of the cylinder, energy transfer in form of heat is improved (see eqn. (1)). In fig. 3b) the same structure is used but with a much deeper cavity ($l_2 \gg l_1$). In this case, the geometric interaction with the rotating cylinder is different. Assuming that the angular velocity of the cylinder is the same in both cases, the velocity gradient will be the same, too. Using the Reynolds analogy, no deviation between both Nusselt numbers is identifiable. However, in fig. 3b) less heat and mass transfer obviously occurs, due to the larger value of l_2 .

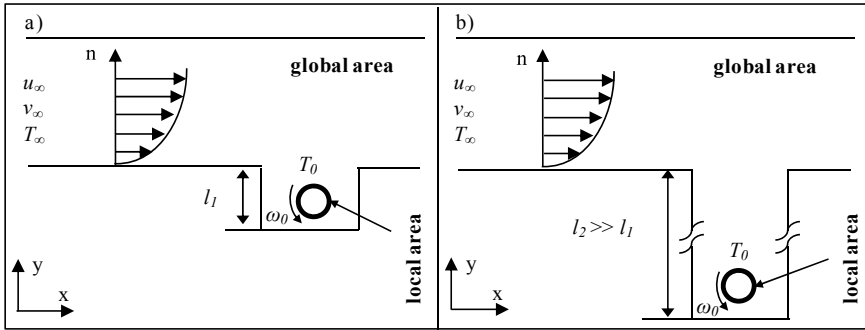


Figure 3: Influence of a design parameter on heat and mass transfer.

2.2 Description of the canonical configuration

To investigate transport phenomena and to distinguish between local and global processes, the authors propose now a new numeric configuration.

It is physically based on the cases described above. The objective is to describe a variety of relevant processes with the same geometrical configuration and involving only a small number of parameters. The retained geometry is shown in fig. 4 and involves a rotating cylinder with a surrounding sleeve that has two openings along the main flow direction.

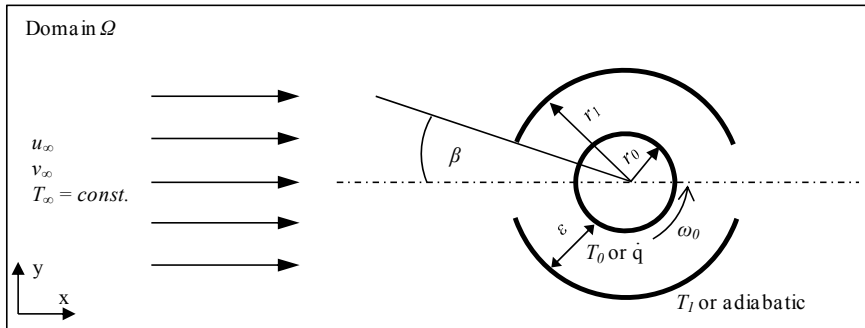


Figure 4: Canonical configuration: Geometry of the rotating cylinder with open sleeve in computational domain Ω .

The opening angle β defines the openings on both sides, equally. Its range is obviously defined with $0 \leq \beta \leq \pi/2$. The gap ε between the cylinder and the external sleeve is strictly positive. The flow is strongly influenced by the surrounding sleeve. Compared to the isolated rotating cylinder, two additional design parameters have been added (β and ε). Table 1 gives an overview of the parameters which have an influence on the flow field locally (within the annuli) as well as globally (in the whole domain Ω).

Table 1: Parameters influencing the flow field.

	locally	globally
dynamic	u_0	u_∞
heat transfer	$\Delta T, u_0$	$\Delta T, u_\infty$
design	ε	β

The ratio of inner (r_0) and outer radius (r_l) is defined as Π . Due to that the gap ε can be computed as $(\Pi-1)r_l$. The dimensionless temperature is $\Phi=(T-T_0)/\Delta T$ with $\Delta T=T_l-T_0$ (assuming that $T_0<T_l$). The dimensionless temperature difference is $\Lambda=\Delta T/T_0$. The Brinkman number (Br) is defined as the ratio between the thermal energy which is produced due to dissipation and the ability of the fluid to evacuate this energy:

$$Br = Ec \, Pr = \frac{(r_0 \omega_0)^2 \mu}{k \Delta T}. \quad (6)$$

The cylinder and sleeve are mounted in a rectangular domain Ω , which is constructed as three-dimensional with symmetry boundary conditions in the cross directions. The flow is considered as steady and laminar. Ideal gas has been chosen as working fluid with the properties of air. Due to the small temperature change in the flow field, the thermo-physical properties such as dynamic viscosity (μ) and thermal conductivity (k) are assumed to be constant. Considering the model description, the physical model becomes a single cylinder at the limit case of $\beta=\pi/2$. Therefore, the characteristic length scale for Re (see eqn. (2)) is defined with $2r_0$ and the characteristic velocity is u_∞ . Considering eqs. (1)...(3), Nu becomes:

$$Nu = (2r_0)^2 \frac{\omega_0}{2u_\infty \varepsilon}. \quad (7)$$

Assuming that $\varepsilon \ll 2r_0$, the velocity profile in the annuli can be considered as linear. When the design parameters r_0 and ε are fixed, only the dynamic parameters have an influence on the Nusselt number. Using this canonical configuration it is possible to analyse separately the influence of u_0 (locally) and u_∞ (globally). Regarding eqn. (2) the Reynolds number is defined as $Re=u_\infty 2r_0/\nu_\infty$. Based on similarity theory, the angular velocity ω_0 can be expressed in a dimensionless form, too.

Table 2: Dimension matrix and relevant quantities.

	h_0	ω_0	u_∞	$2r_0 \sin \beta$	ρ	c_p	$2r_0$	k	ν_∞
mass	1	0	0	0	1	0	0	1	0
length	0	0	1	1	-3	2	1	1	2
temp.	-3	-1	-1	0	0	-2	0	-3	-1
time	-1	0	0	0	0	-1	0	-1	0

Table 2 shows the corresponding dimension matrix with the relevant quantities characterizing the physical model shown in fig. 4. After solving the equations, the dimensionless tangential velocity (I) can be defined as:

$$\Gamma = \frac{\omega_0(2r_0)^2}{\nu}. \quad (8)$$

3 Numerical method

All the equations are discretised with a finite-volume formulation relying on the Reynolds-Averaged Navier-Stokes (RANS) approach with a second-order upwind scheme. For the mesh, hexahedra elements are chosen. For the calculation of Nu , a reference temperature (T_{Bulk}) is needed. As suggested in Zhang *et al.* [21] T_{Bulk} was defined with the ambient temperature T_∞ . The viscous work term was included in the energy equation, since it is needed for calculations with high velocity gradients. The convergence criteria for mass, velocity components and enthalpy is set as a factor of 10^{-6} . There was no difference in the convergence behaviour with including viscous work term.

4 Grid-independence test and code validation

In this section, the dependency of the grid on a semi-infinite flat plate is tested, as well as the implemented numerical code on two concentric, rotating cylinders.

4.1 External flow: semi-finite plate

Herwig and Schmandt [5] considered the drag coefficient (c_D) for an infinite plate as follows:

$$c_D = 2[1.328Re^{-1/2} + 2.67Re^{-7/8}]. \quad (9)$$

For an alternative formulation of c_D , they suggested to use the SLA-approach (entropy generation occurring due to fluid friction only) for the calculation. Based on the Gouy-Stodola-Theorem c_D can be expressed as:

$$c_D = \frac{T}{\frac{\rho}{2}u_\infty^3 A} S_F'''. \quad (10)$$

where S_F''' is the entropy generation due to fluid friction. Fig. 5 compares the results of eqn. (10) with eqn. (9), showing the excellent agreement between both results in the relevant range for Re .

A grid-independence test was performed for the semi-infinite plate. For a laminar flow, the results are matching very well with each other, under the condition that $y^+ \leq 1$. For the first mesh, $y^+ > 1$ is observed for $Re > 4096$. Around this Reynolds number the red dotted line starts diverging from the expected solution and leads to increasingly wrong results, strongly underestimating c_D . The same effect appears for the second mesh around $Re = 100\,000$ (dashed line). The third and finest mesh show a good agreement with the analytical results till $Re > 500,000$, since $y^+ < 1$ always apply. These results corroborate the recommendations of ANSYS [22], since $y^+ < 1$ is recommended for high accuracy.

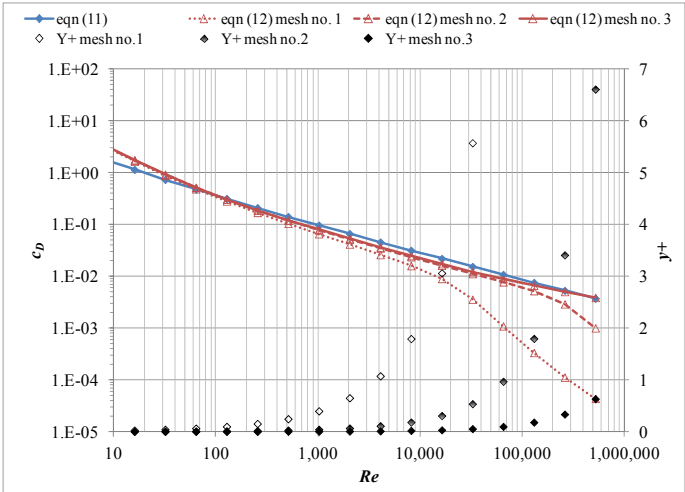


Figure 5: Drag coefficient for the semi-infinite flat plate.

4.2 Internal flow: concentric rotating cylinders

In a further study, the flow between two concentric rotating cylinders has been investigated as an example for an internal flow (corresponding to the local flow between the two cylinders for our canonical configuration at $\beta=0^\circ$). The ratio between the outer and inner tangential velocity is defined as $\lambda=\omega_1r_1/(\omega_0r_0)$. The dimensionless radial distance is $R=r/r_0$. Fig. 6 shows the numerically calculated velocity profile compared to the analytical solutions from Mahmud and Fraser [14] for a varying λ , with $U=u/(\omega_0r_0)$.

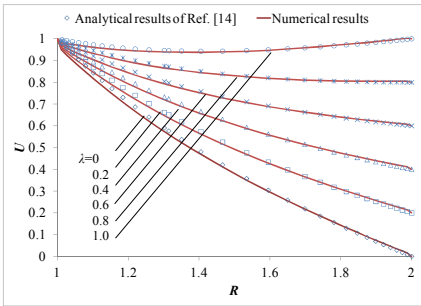


Figure 6: Effect of λ on U for $\beta=0^\circ$, $\Pi=2$ and $Br=0$.

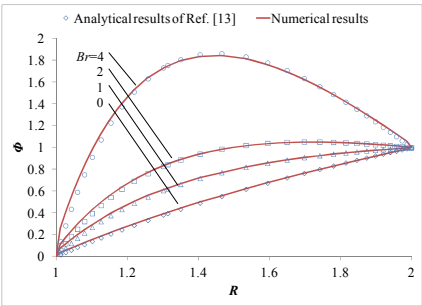


Figure 7: Effect of Br on Φ for $\beta=0^\circ$, $\Pi=2$ and $\lambda=0$.

The numerical solutions are in excellent agreement with the analytical solution for the cases with $0\leq\lambda\leq1$, when the tangential velocity of the inner cylinder is constant and the outer tangential velocity is changed in magnitude.



Table 3 shows the y^+ values employed for each study. As appears from the table, the y^+ values are much smaller than 1.

Table 3: Y^+ values at the inner and outer cylinder wall and Reynolds numbers in the gap for different values of λ .

	$\lambda=0$	$\lambda=0.2$	$\lambda=0.4$	$\lambda=0.6$	$\lambda=0.8$	$\lambda=1.0$
y^+ inner cylinder	0.088	0.084	0.079	0.074	0.068	0.062
y^+ outer cylinder	0.034	0.042	0.040	0.037	0.034	0.031
Re in gap	16.4	20.2	23.9	27.7	31.5	35.3

Fig. 7 shows the numerical solutions for the dimensionless temperature profile Φ in comparison to the analytical solution from Mirzazadeh *et al.* [13]. Both cylinders are associated with an isothermal boundary condition. The results show, that the analytical solutions match almost perfectly with the numerical results. To vary Br for the temperature profiles, the dynamic viscosity μ was changed to $1.83 \times 10^{-2} \text{ kg m}^{-1} \text{ s}^{-1}$ for this study. Because of $\lambda=0$, y^+ values were calculated for the inner cylinder, only (table 4).

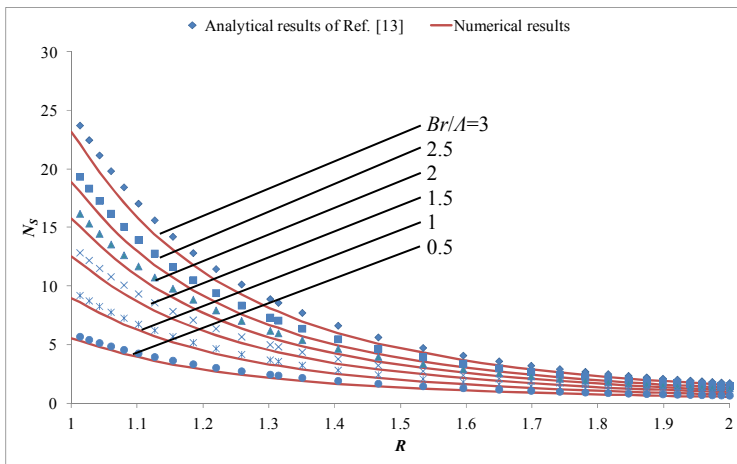


Figure 8: Effect of Br/A on N_s for $\beta=0^\circ$, $It=2$ and $\lambda=0$.

Table 4: Y^+ values at the inner cylinder for different values of Br .

	$Br=0$	$Br=1$	$Br=2$	$Br=4$
y^+ inner cylinder	0.003	0.01	0.012	0.016
Re in gap	0.01	0.2	0.3	0.5

The entropy generation (S_{gen}''') can be obtained by (see Bejan [3]):

$$S_{gen}''' = S_C''' + S_F''' = \frac{k}{T^2} (|\nabla T|)^2 + \frac{\mu}{T} \left(\frac{\partial U_i}{\partial x_j} + \frac{\partial U_j}{\partial x_i} \right) \frac{\partial U_i}{\partial x_j}. \quad (11)$$

The associated dimensionless form N_S is equal to the ratio of S_{gen}''' to a characteristic entropy generation rate ($S_{G,C}$). These are for both thermal boundary conditions:

$$S_{G,C} = \left[\frac{q^2}{kT_0^2} \right]_{\text{Isoflux}} \quad S_{G,C} = \left[\frac{k(\Delta T)^2}{r_0^2 T_0^2} \right]_{\text{Isothermal}} \quad (12)$$

Fig. 8 shows the dimensionless entropy generation number N_S for a varying Br/A ratio, compared to the analytical results from Mirzazadeh *et al* [13]. The highest relative derivation between theory and numeric is below 8% and is found for $Br/A=3$. Concerning the y^+ and Re values, the results in table 3 can be used.

As a whole, the validation of our numerical approach is successful. For most tested configurations and parameters, the numerical results are identical or nearly identical to the analytical prediction. Small discrepancies, below 8% are only observed for the internal flow between two concentric cylinders. An explanation for these larger differences could not be found yet. Value of y^+ well below 1 are needed for an accurate analysis of flow, heat transfer, and associated entropy generation.

5 Results and discussion

Applying now the developed approach for a first study, the design parameters for our canonical configuration were set to $\beta=30^\circ$ and $\Pi=2$ with an adiabatic sleeve. Fig. 9 shows the resulting Nusselt number for varying angular velocities ω_0 (Γ) and inlet velocities u_∞ (Re).

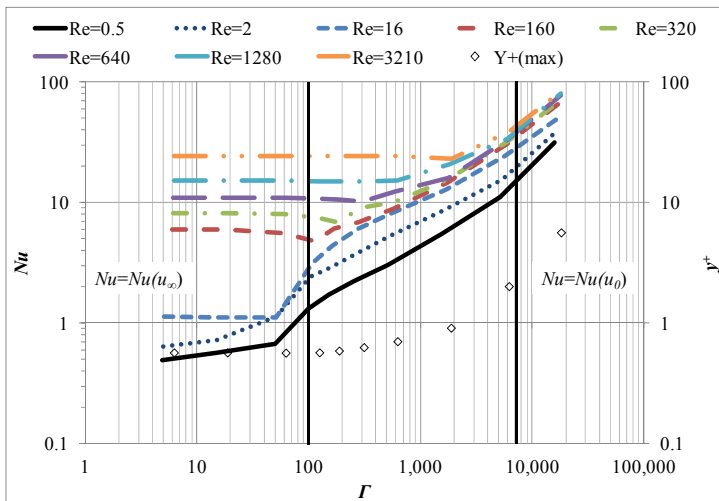


Figure 9: Effect of design parameters on Nu for $\beta=30^\circ$, $\Pi=2$ and $Br=0$.

When increasing Γ the effect of the tangential speed of the cylinder ($u_0 = \omega_0 r_0$) overweighs the inlet velocity u_∞ . In the range $1 \leq \Gamma \leq 6000$ the Reynolds number is the dominating effect for Nu . For the range $\Gamma > 6000$ the heat and mass transfer mainly depend on the angular velocity ω_0 . The y^+ values used in the simulation increase linearly with the magnitude of Nu , so that the expected accuracy is lower in the right part of the diagram.

6 Conclusions

The first and second laws of thermodynamics have been used in combination with numerical simulations to investigate different configurations. A canonical configuration has been proposed based on fundamental geometries used for aerodynamic studies. This canonical configuration allows studying different transport phenomena correlated to different types of dynamic, heat transfer and design parameters. Numerical studies for external and internal flows were done to check grid-independence and for the purpose of code validation. Based on these first results, deriving suitable indicators based on entropy generation in the canonical configuration appears to be possible. As a first application example, the Nusselt number was calculated for $\beta = 30^\circ$, $\Pi = 2$ and $Br = 0$. For the defined boundary conditions the transition area for Nu is between $100 \leq \Gamma \leq 6000$. At values of $\Gamma \leq 100$ the inlet velocity u_∞ is the dominating effect. At higher values of Γ the influence of the tangential velocity u_0 overweighs u_∞ . Further studies will systematically vary the corresponding parameters in order to derive corresponding laws.

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